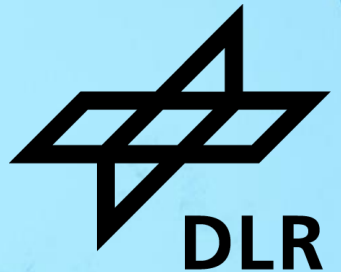


SENSING THE PULSE OF THE EARTH WITH QUANTUM TECHNOLOGY

Prof. Dr. Matthias Weigelt



What is the pulse of the Earth?



Water cycle



Ocean circulation



Carbon cycle



Nutrient cycle

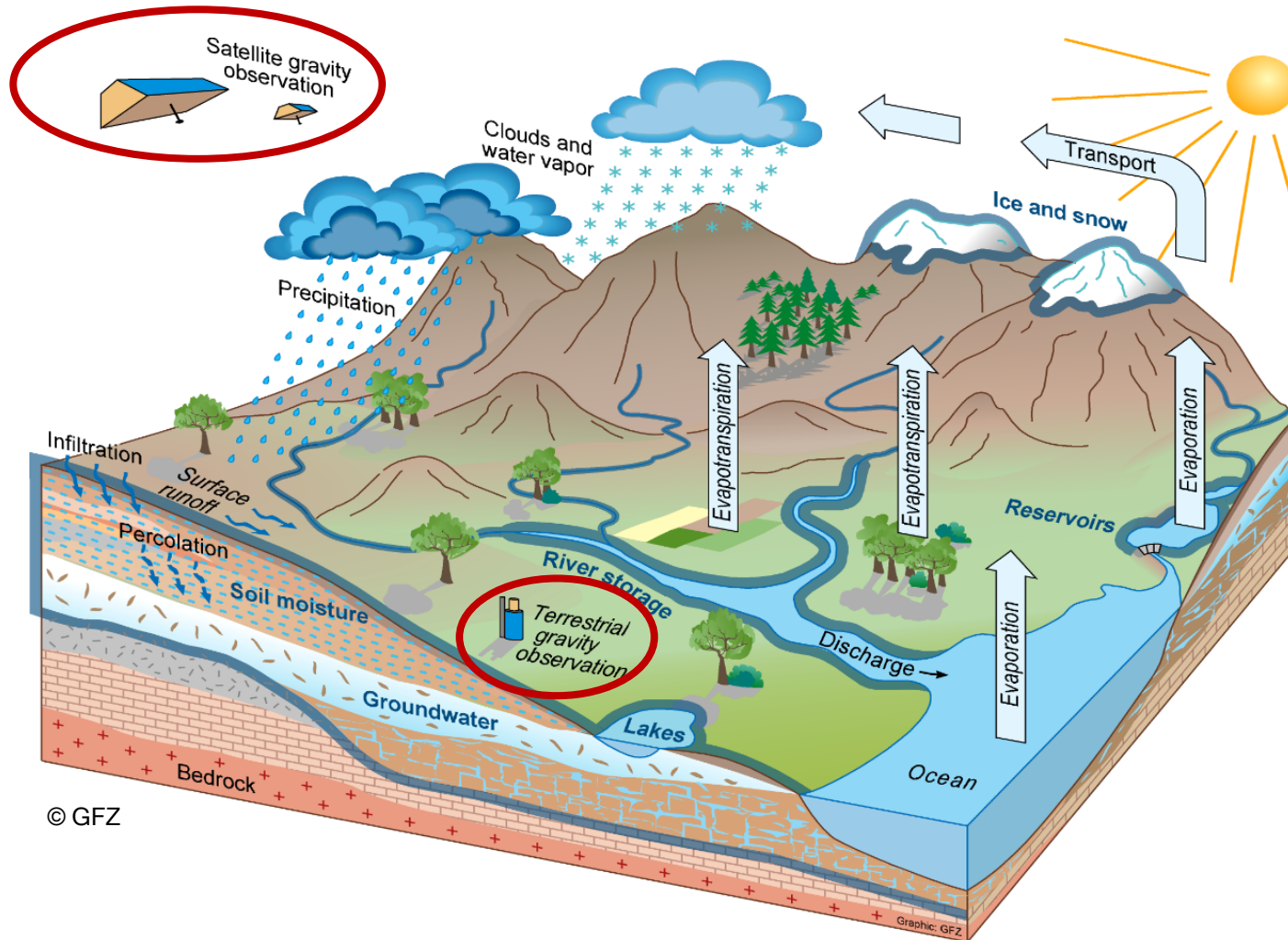


Rock cycle



Gravimetry is a universal tool to sense the pulse of the Earth

Water cycle



© GFZ

Groundwater Soil moisture Lakes River storage Ice and snow Reservoirs

↓

TOTAL WATER STORAGE



Continental water balance

$$\Delta S = P - ET - Q$$

- P: Precipitation
- ET: Evapo(trans)piration
- Q: Runoff
- ΔS : Storage change

The carbon cycle in sight ...

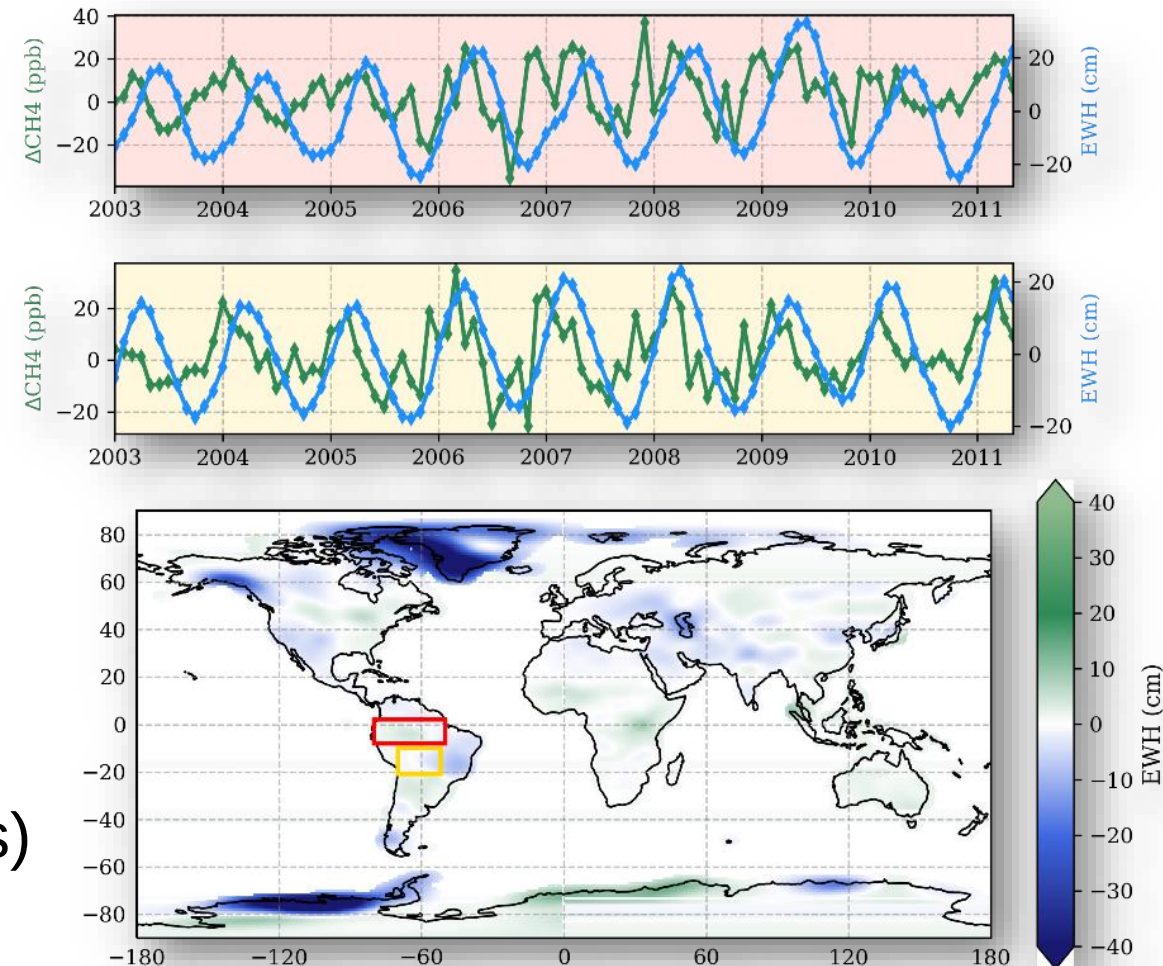
Wetlands are the largest natural Methan (CH_4) source

Project:

*Wechselwirkung zwischen
Kohlenstoff- und Wasserkreislauf*



CH_4 is produced during
anaerobic decomposition (Methanogenesis)



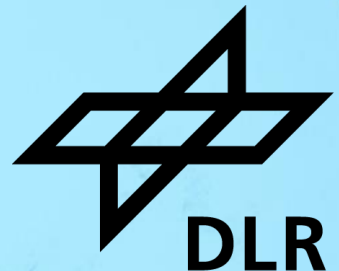


Observing gravity is observing climate change

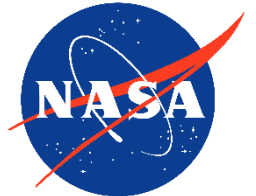


INSTITUTE FOR SATELLITE GEODESY AND INERTIAL SENSING

Deutsches Zentrum für Luft und Raumfahrt (DLR e.V.)
Institut für Satellitengeodäsie und Inertialsensorik (DLR-SI)
Callinstrasse 30b, 30167 Hannover



Our mission



AIRBUS



...

DLR-SO

DLR-GK

DLR-SC

DLR-FL

DLR-QT

DLR-KN

DLR-RY

...

Satellite geodesy

- highly precise data for climate change
- better gravity field data
- global reference systems for space and time using ultra-cold atoms

Inertial sensors

- navigation for intelligent mobility
- new sensors for satellite missions and exploration
- Quantum gas und optical interferometry

Research

- mission and concept studies
- data analysis
- simulations
- hybrid sensor technology

Development

- space qualification
- prototypes
- systems engineering tools

Technology transfer

- cooperationen (inter-)national
- industry
- start-up
- patents

DLR-SI in numbers



- Locations
 - Hannover
 - Bremen

- Human Resources
 - Scientific Personal 40
 - HiWi 4
 - Master/Bachelor 3
 - Administrative MA 6

- Close co-location to the Universities

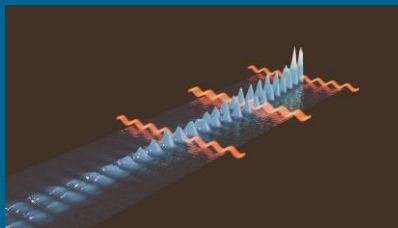
Utilizing the fundamental properties of atoms and molecules
to improve measurement accuracy and resolution

Laser interferometry



- Long-term stability, high-precision distance measurements
- High-precision measurement of the smallest angles

Atom interferometry



- Absolute, drift-free, highly accurate
- Accelerometer
- Rotation sensor

Optical clocks



- absolute, drift-free, highly accurate
- navigation
- reference system for height profiles

Digital Twin: orbit propagation considering all disturbing forces (environment, space weather, satellite), simulation of sensors, systems engineering tools

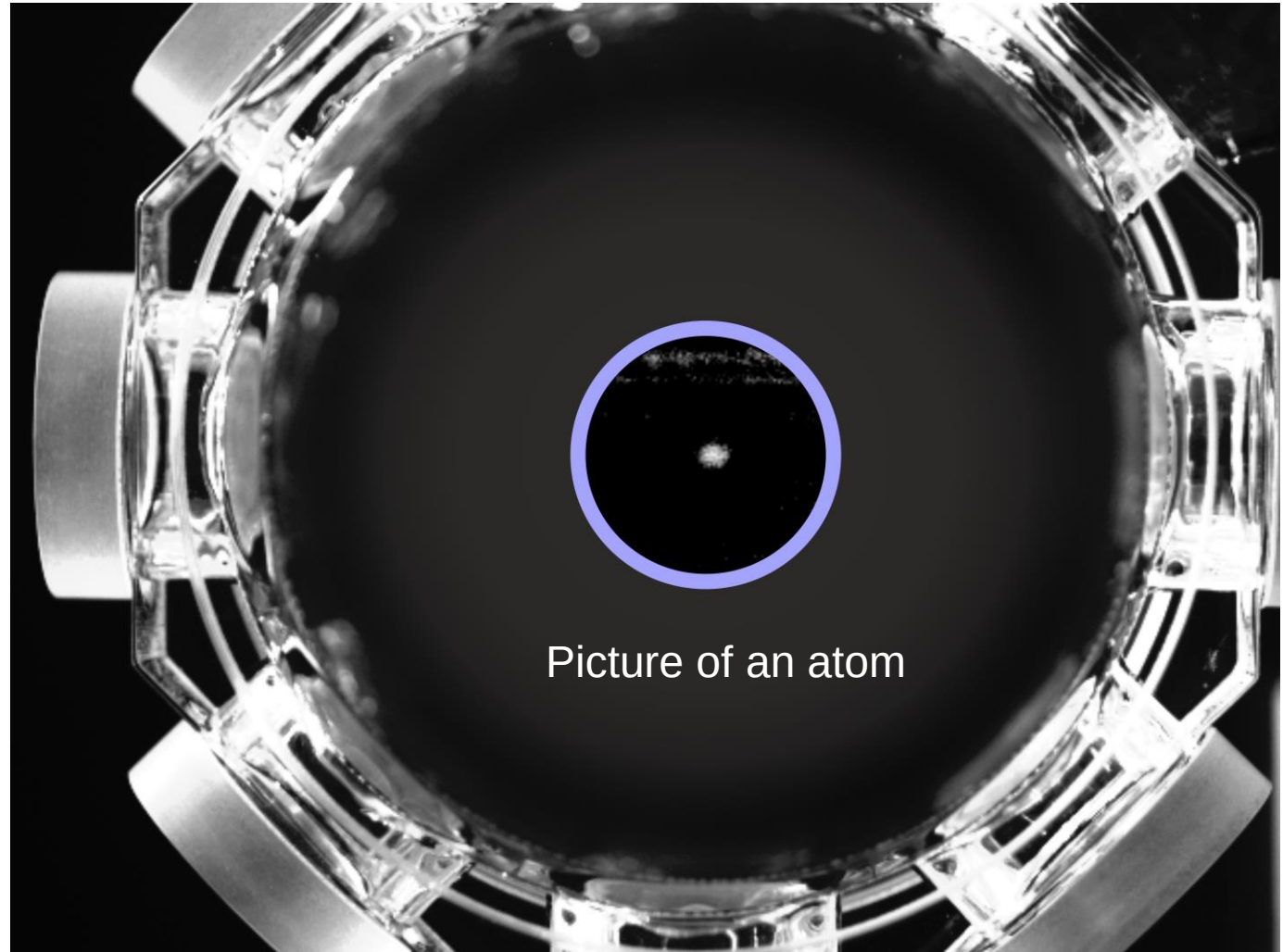
Geodetic Modelling

Preparation of gravity field data for the public and data analysis tools

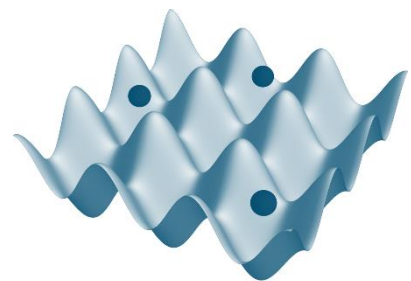
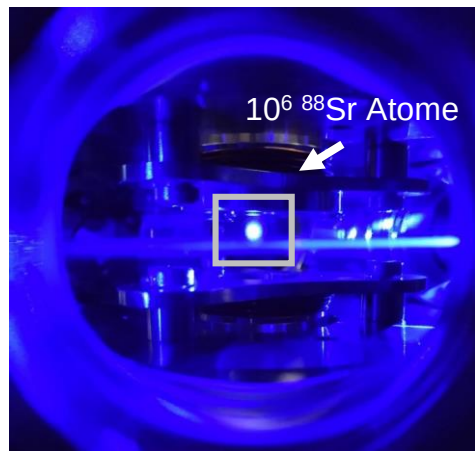
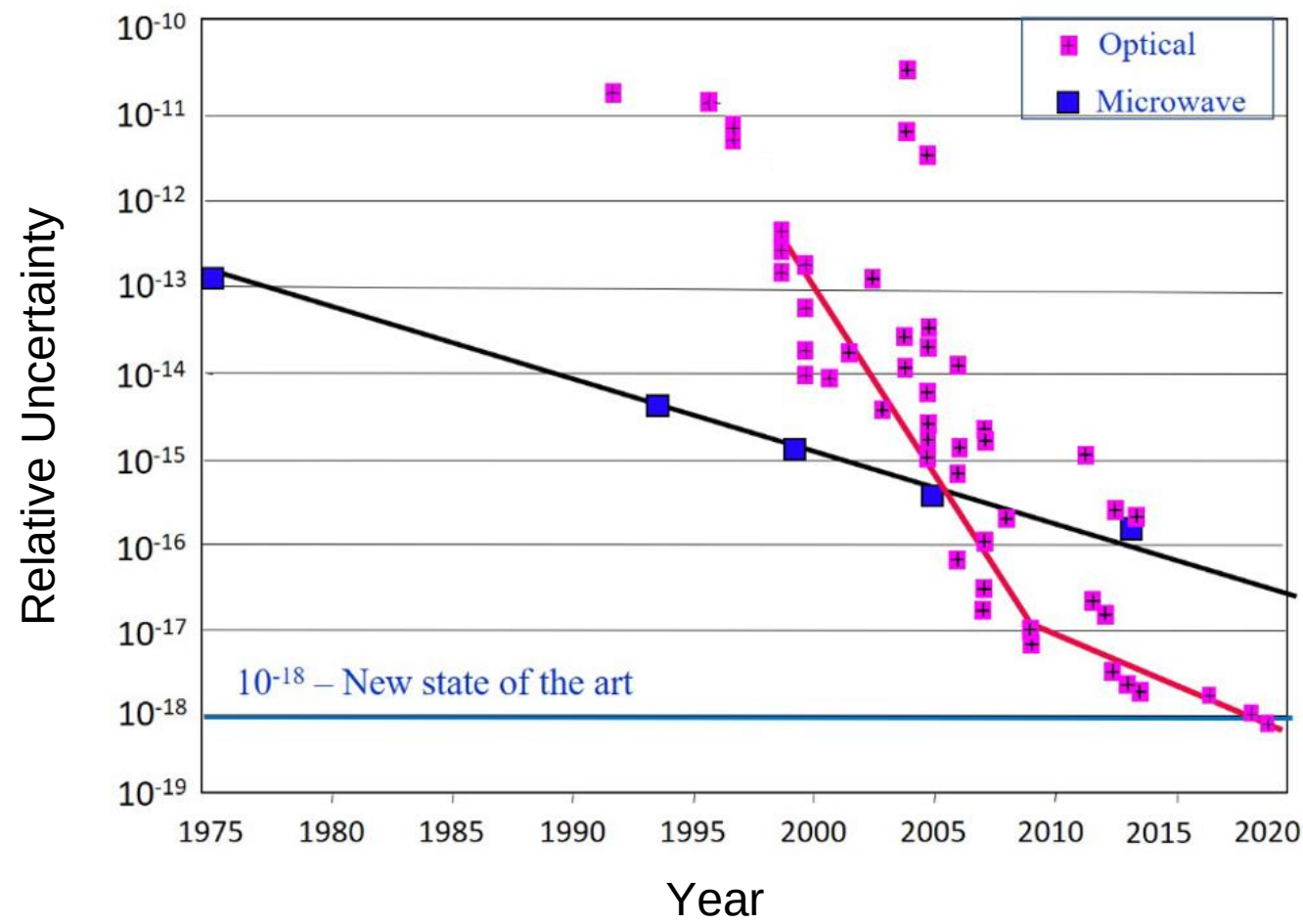
OPTICAL FREQUENCY METROLOGY

Quantum Sensors

- Isolating atoms from a noisy environment
- Controlling atoms on the quantum level
(→ wave properties)
- Using atoms for ultra-precise interferometer
 - Accelerometry
 - Rotational Sensors
 - Gravimetry/Gradiometry
 - Time- and Frequency



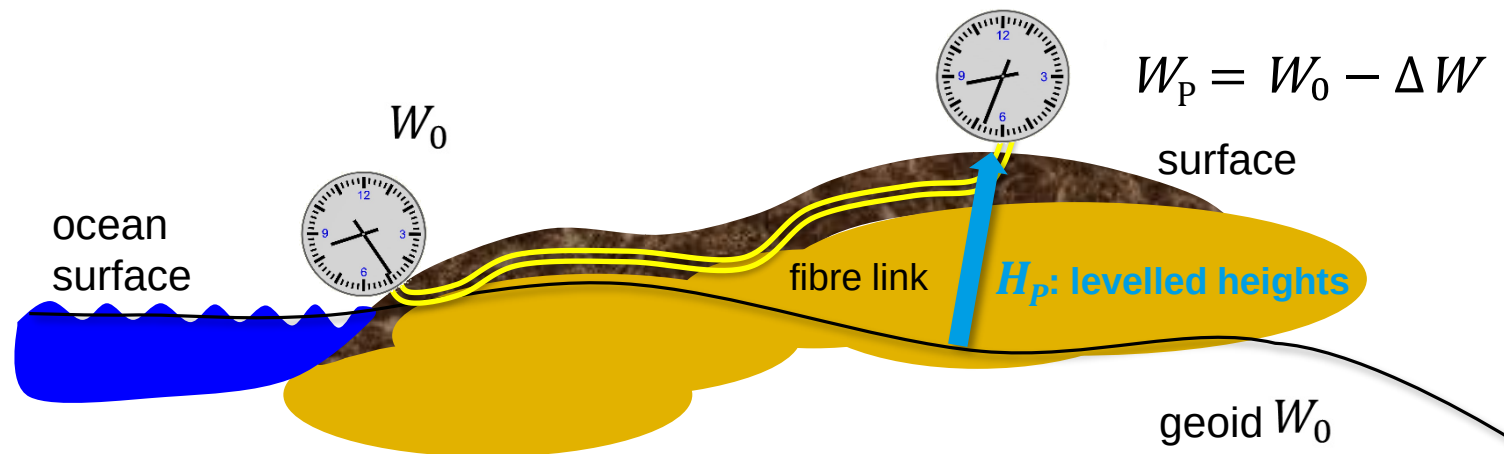
Improving the precision of clocks



Optical lattice clock

Networks of Clocks: Objectives

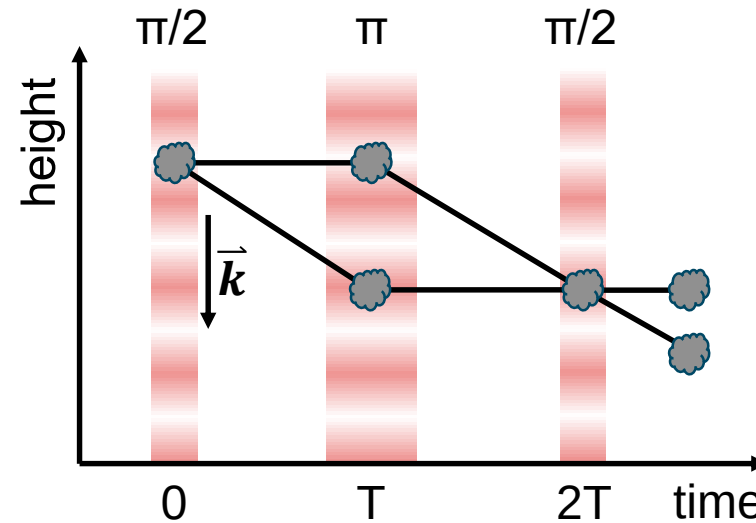
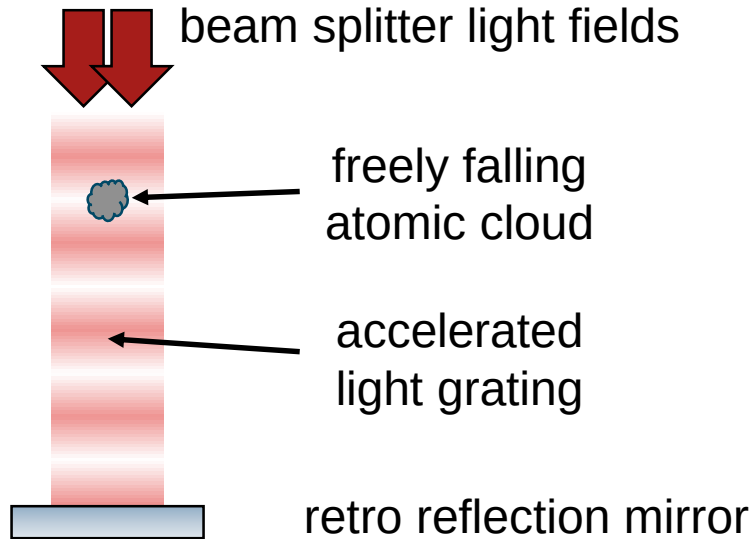
- relativistic frequency change: $\frac{\Delta f}{f} = \frac{W_0 - W_P}{c^2}$
 - gravity potential W : Newtonian + centrifugal terms
 - height: $H_P = \frac{W_0 - W_P}{\bar{g}} = \frac{c^2}{\bar{g}} \frac{\Delta f}{f}$
- ➔ Chronometric levelling over long distances
- ➔ Clock-based height system



ATOM INTERFEROMETRY

Atom interferometry

Atoms in free fall, Mach-Zehnder like $\pi/2 - \pi - \pi/2$ pulse geometry:



Interferometer phase, ϕ_i imprinted at pulse i:

Acceleration $\vec{a}$, effective wave vector $\vec{k}$:

Forward drift velocity $\vec{v}$, rotation $\vec{\Omega}$, enclosed area $\vec{A}$:

$$\phi_{tot} = \phi_1 - 2\phi_2 + \phi_3$$

$$\phi_{acc} = \vec{k} \cdot \vec{a} T^2$$

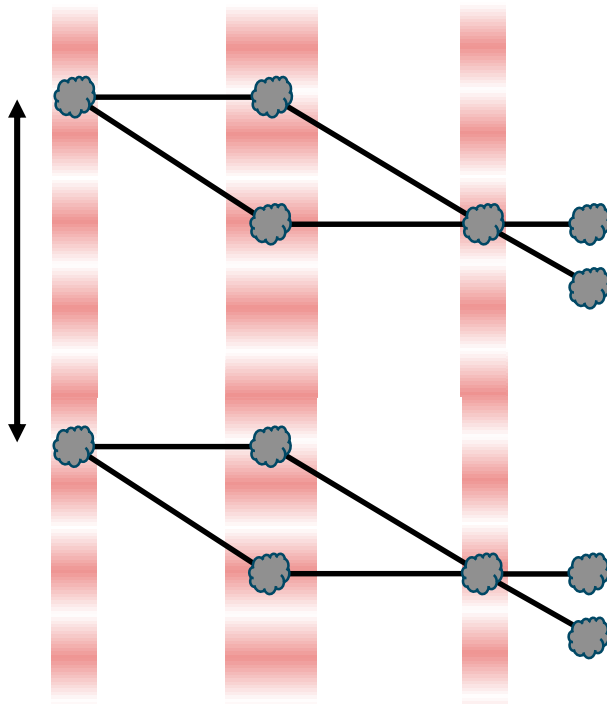
$$\phi_{rot} = 2(\vec{k} \times \vec{v}) \cdot \vec{\Omega} T^2 \sim \vec{A} \cdot \vec{\Omega}$$

Gravity gradiometer

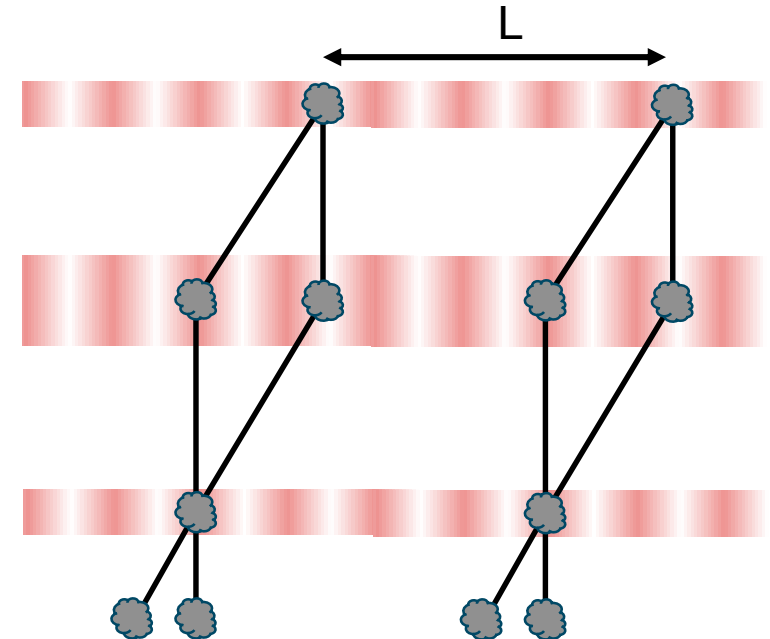
Combination of two Mach-Zehnder like atom interferometers, simultaneous interrogation with the same light field → extraction of mean gravity gradient between the two interferometers from the differential signal

Vertical
orientation

Baseline
separation L



Horizontal
orientation



[McGuirk et al., PRA 65, 033608 (2002); Fixler et al., Science 315, 74 (2007); Biedermann et al., PRA 91, 033629 (2015); Chiow et al., PRA 93, PRA 93, 013602 (2016); Rosi et al., Nature 510, 518 (2014); Asenbaum et al. PRL 118, 183602 (2017)]

State of the art in AI-based quantum sensors



Rotation sensors

Stability:

- 30 nrad/s in 1s
- 0.1 nrad/s after averaging

Uncertainty:

- Few nrad/s to 10 nrad/s

Gravimeters

Stability:

- 42 nm/s² in 1s
- 0.5 nm/s² after averaging

Systematic uncertainty:

- 40 nm/s²
- Limited by wave front distortions ¹⁾

Transportable, sea, flight,
commercial versions

Gravity gradiometers

Stability:

- $3 \cdot 10^{-8}$ 1/s² in 1s

Systematic uncertainty

- $8 \cdot 10^{-8}$ 1/s²

Determination of gravitational
constant

[From: Chen et al., arXiv:2303.00239; Gautier et al., Sci. Adv. 8, eabn8009 (2022); Berg et al., PRL 114, 063002 (2015); Stockton et al., PRL 107, 133001 (2011); Gauguier et al., PRA 80, 063604 (2009); Gillot et al., Metrologia 51, L15-L17 (2014); 1) reduced in Karcher et al., NJP 20, 113041 (2018); Freier et al., JoP:CS 723, 012050 (2016); Hu et al., PRA88, 043610 (2013); Wu et al., Sci. Adv. 5, eaax0800 (2019); Bidel et al., Nat.Comm. 9, 2041 (2018); Bidel et al., JoG 94, 1432 (2020); muquans.com; McGuirk et al., PRA 65, 033608 (2002); Fixler et al., Science 315, 74 (2007); Biedermann et al., PRA 91, 033629 (2015); Chiow et al., PRA 93, PRA 93, 013602 (2016); Rosi et al., Nature 510, 518 (2014); Asenbaum et al. PRL 118, 183602 (2017)]

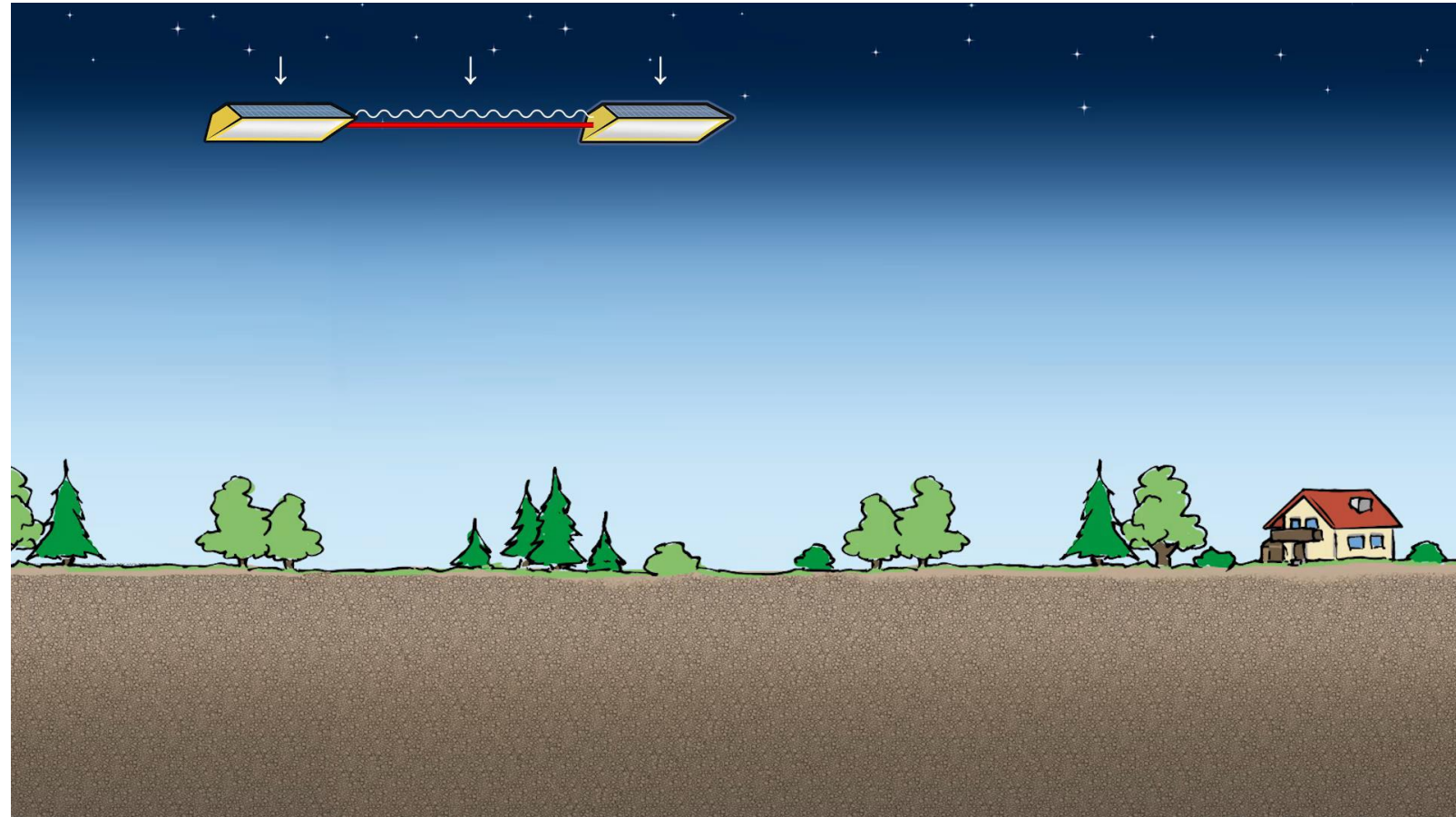
A blurred photograph of a laser interferometry experiment. A bright red laser beam is visible, reflecting off a mirror and passing through optical components. The background is out of focus, showing laboratory equipment and a person's hand.

LASER INTERFEROMETRY

From observing geometry to understanding the forces ...

GRACE-FO observation systems

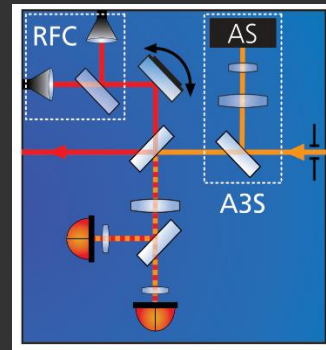
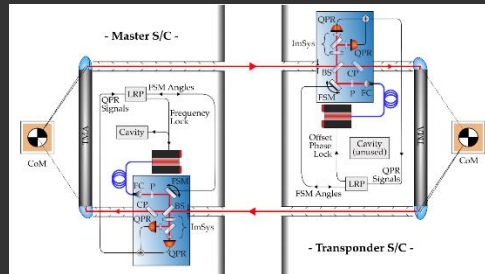
- Initial orbit height: 485 km
- Inclination: $\sim 89^\circ$
- Key technologies:
 - GPS
 - Accelerometer
 - Ranging system
- Observation quantity:
 - distance (range)
 - change of distance (range rate)



Satellite Gravimetry with laser interferometry

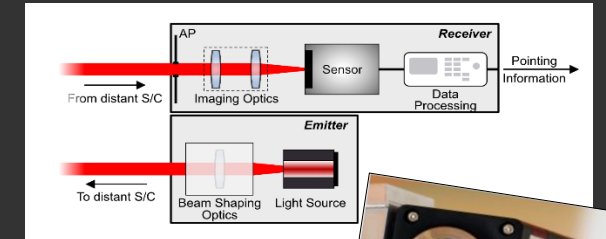
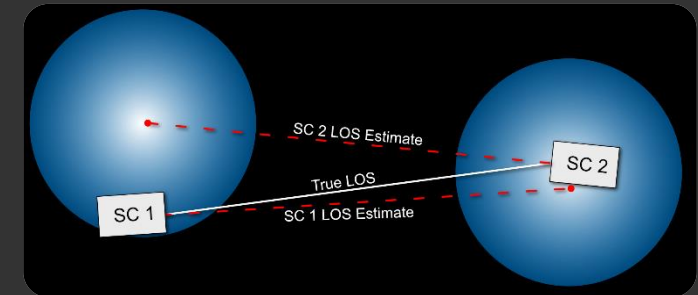
Components for the next generation optical bench designs

- Laser ranging as main science instrument requires improved redundancy schemes
- Miniaturization of components
- Novel concepts



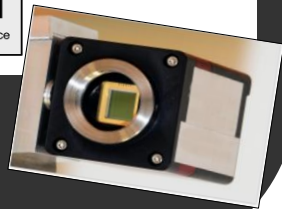
Dedicated Constellation Acquisition Systems

- Laser link acquisition is the most critical step during commissioning of the instrument
- Dedicated hardware mitigates risks and speeds up this process

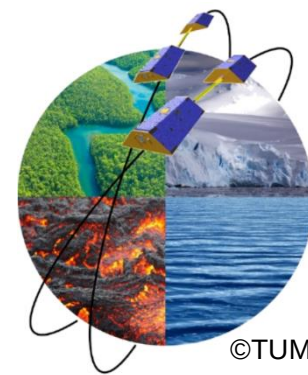
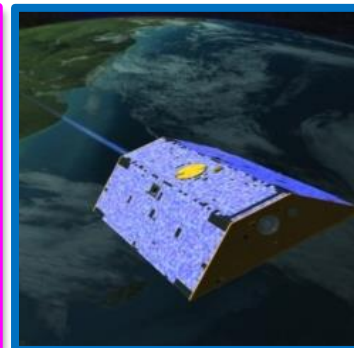
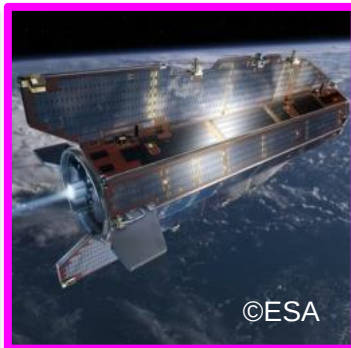
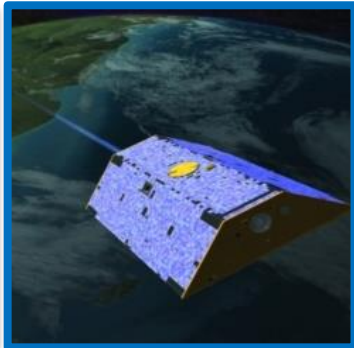
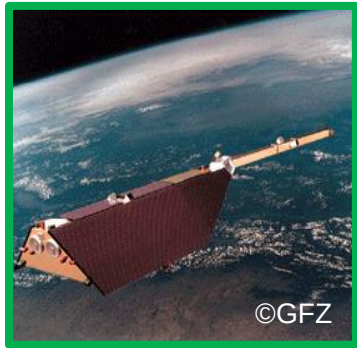


MiniCAS:

- Compact, modular, low SWaP design



Continuity of mass transport measurements



CHAMP (GFZ, 2000-2010)

GRACE (NASA/DLR, 2002-2017)

GRACE-FO (NASA/GFZ, 2018-2030)

GOCE (ESA, 2009-2013)

GRACE-C (2028-203?)

MAGIC

NGGM (2031-203?)

Gravity missions enabled spectacular results:

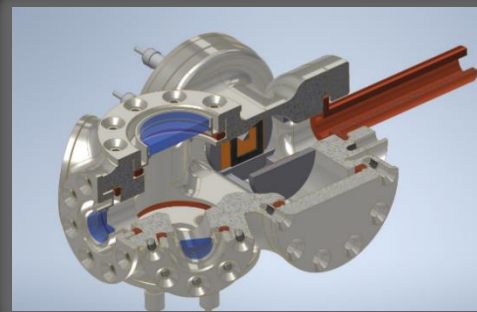
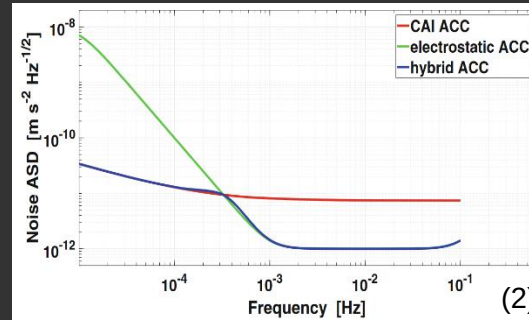
- insights into the global water cycle
- polar and mountain ice mass loss
- changes in ocean surface currents
- unification of height systems
- sea level rise

Satellite Gravimetry in the post-MAGIC era

- Accelerometry -

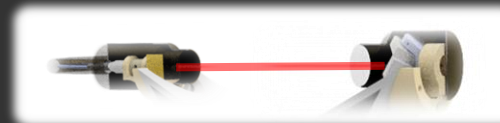
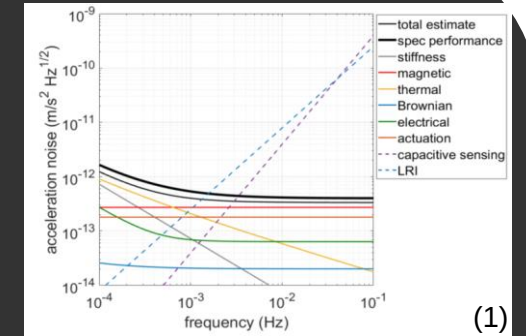
A) Cold Atom Interferometers

- Hybridization with optical sensor
- Absolute measurements
- Drift-free
- Superb precision



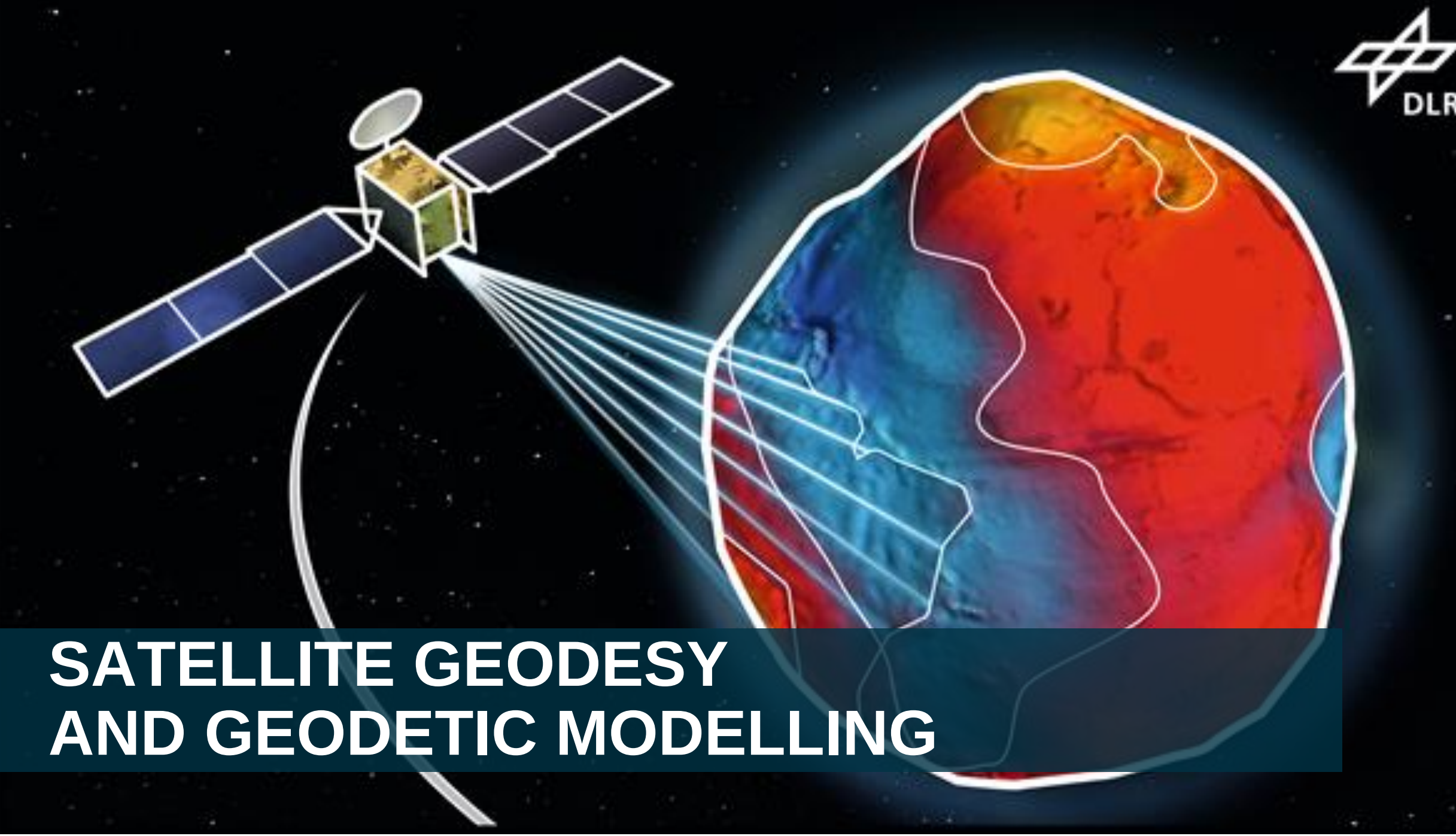
B) Gravitational Reference Sensor

- Based on heritage from LISA Pathfinder
- Superb precision
- Long-term stable
- Capacitive or optical readout
- Multi-axis optical readout requires compact interferometric setups



(1) <https://doi.org/10.1007/s00190-022-01659-0>

(2) https://doi.org/10.1007/1345_2022_151

The background of the slide is a dark space scene. On the left, a satellite with a central body and two long solar panel arrays is shown. It is emitting a fan of white lines representing its field of view or data collection area. These lines converge on a portion of the Earth on the right. The Earth is depicted with a color-coded surface, where red and orange hues likely represent higher elevations or temperatures, and blue hues represent lower elevations or cooler temperatures. The satellite's orbit is indicated by a thin white arc around the Earth.

SATELLITE GEODESY AND GEODETIC MODELLING

ATOM INTERFEROMETRY IN SPACE

CARIOQA-PMP

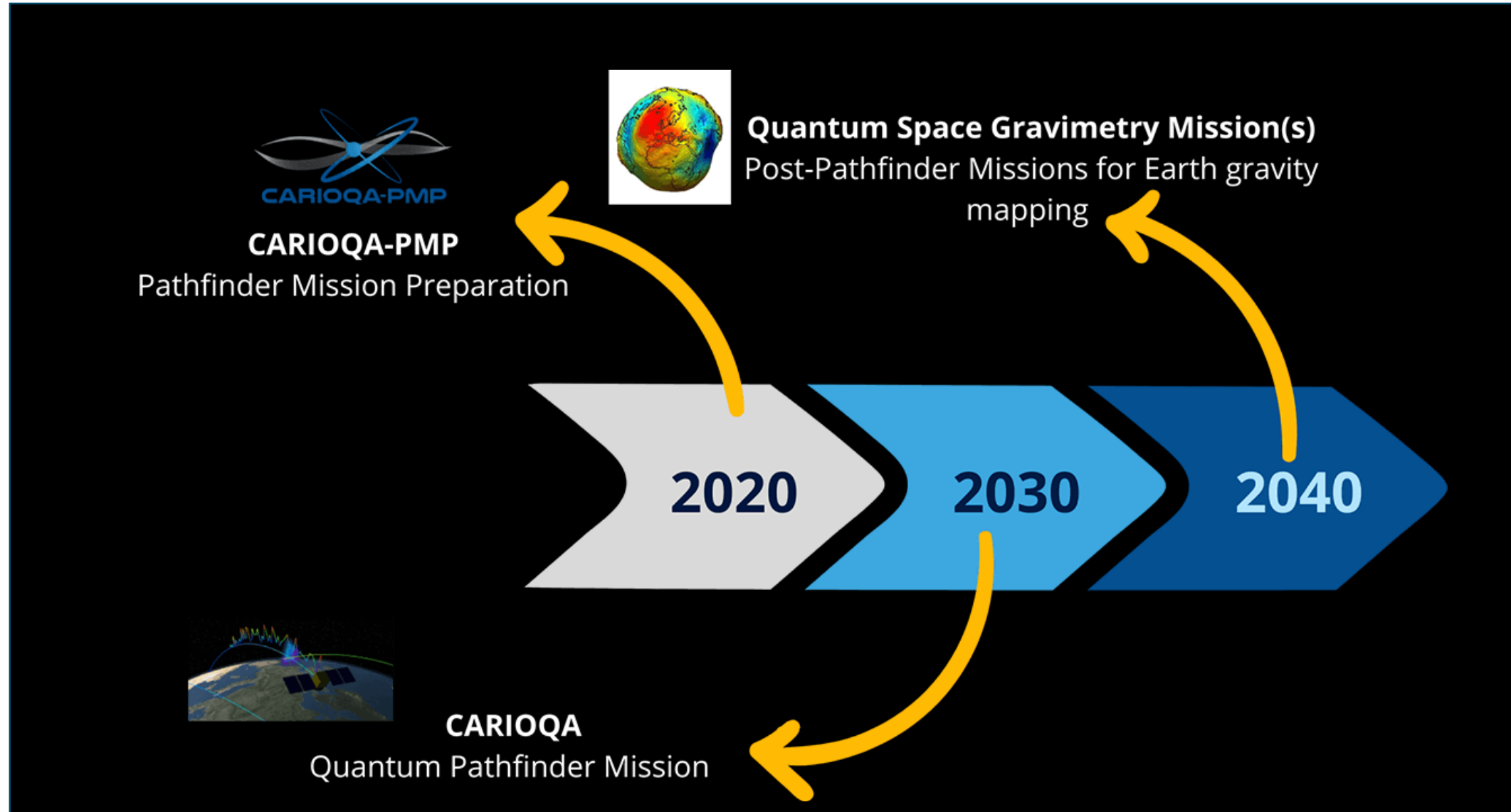
stands for

Cold **A**tomium **R**ubidium **I**nterferometer in **O**rbit for **Q**uantum **A**ccelerometry
- **P**athfinder **M**ission **P**reparation -

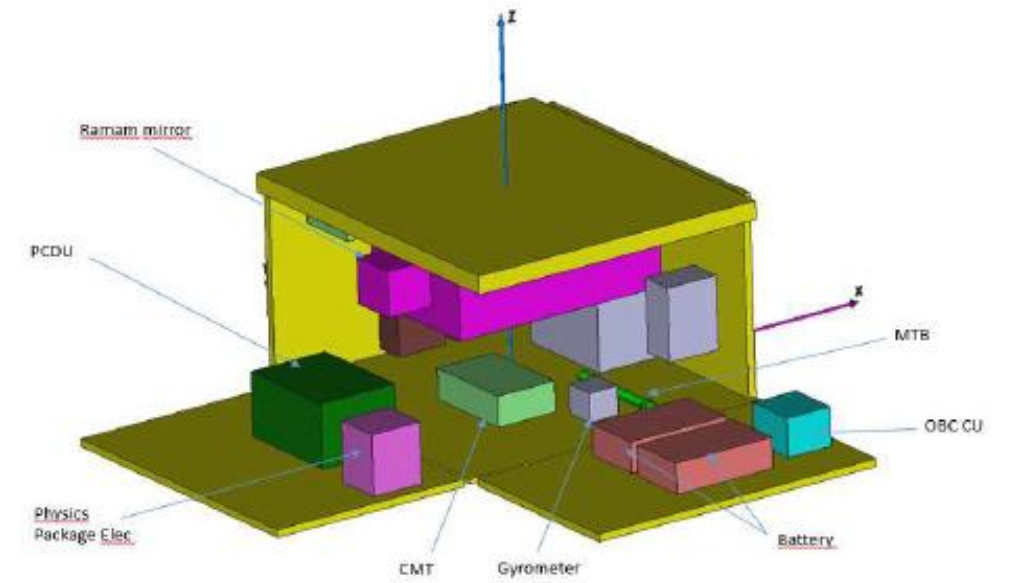
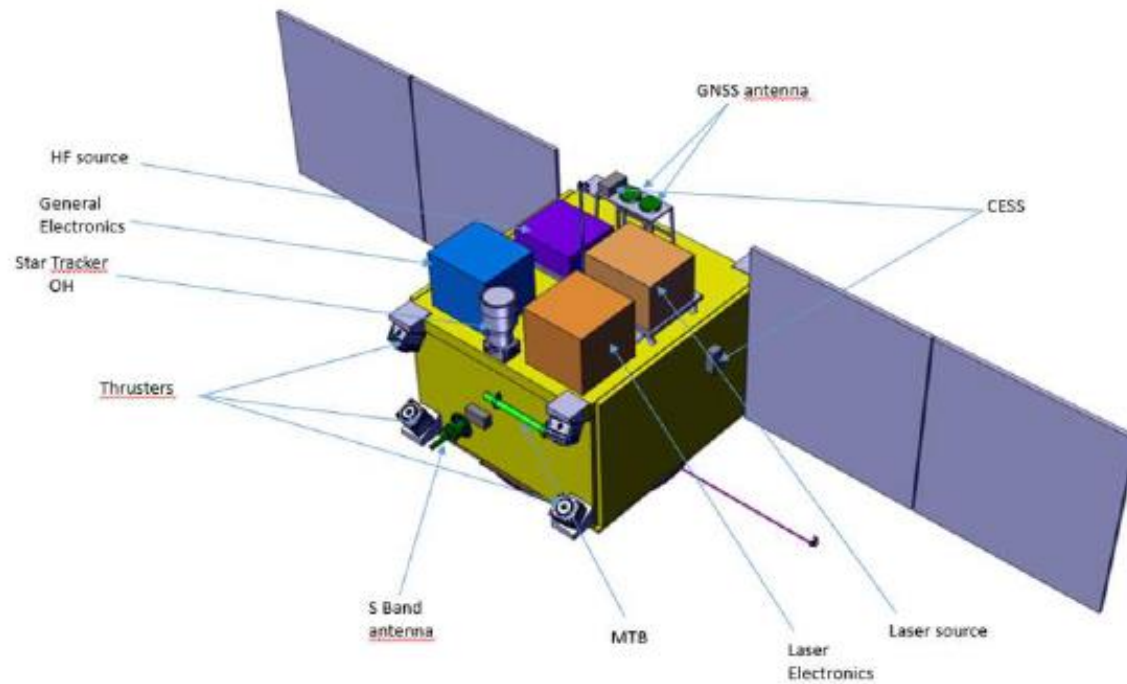


Through a Quantum Space Gravimetry Pathfinder Mission
CARIOQA-PMP aims at developing a new technology to be used in space
within the next decade: a quantum gravimeter/ accelerometer

Timeline



Preliminary satellite concept



from Lévêque et al. 2022

CARIOQA-PMP brings together **leading 17 players** from **5 EU countries**:



Coordination
CNES and DLR under CNES lead



Satellite instrument development

Airbus Defence and Space, Exail, TELETEL, LEONARDO



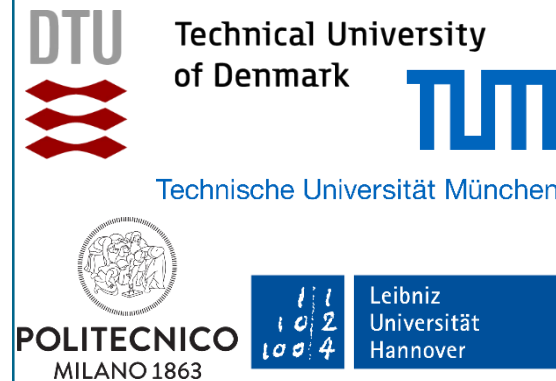
Quantum sensing

LUH, SYRTE, LP2N, LCAR, ONERA, IESL/FORTH



Space geodesy, Earth sciences

LUH, TUM, POLIMI, DTU



Impact maximization and impact assessment

FORTH/ PRAXI Network, G.A.C. Group



Anticipated sensitivity



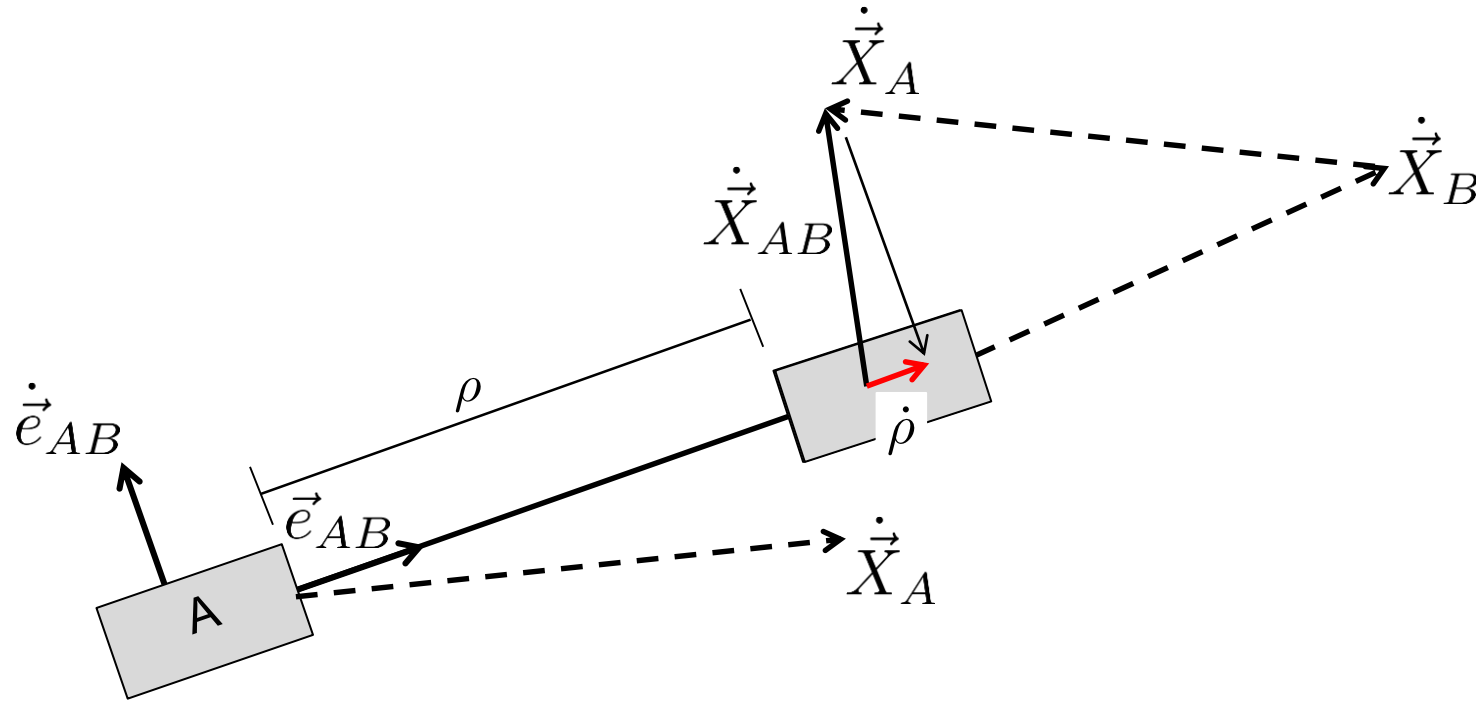
Instrument	Applications	Sensitivity
Absolute quantum gravimeter (SYRTE)	Ground/Laboratory	$5 \times 10^{-8} \text{ m.s}^{-2}.\text{Hz}^{-1/2}$
Commercial absolute quantum gravimeter (IXBLUE/Muquans)	Ground/Field applications	$5 \times 10^{-7} \text{ m.s}^{-2}.\text{Hz}^{-1/2}$
Absolute atom accelerometer/gravimeter in microgravity (ICE/LP2N)	Microgravity	$6 \times 10^{-7} \text{ m.s}^{-2}.\text{Hz}^{-1/2}$
Quantum Pathfinder Mission (CARIOQA)	Satellite	$1 \times 10^{-10} \text{ m.s}^{-2}.\text{Hz}^{-1/2*}$
Quantum Space Gravimetry Missions	Satellite	$1 \times 10^{-12} \text{ m.s}^{-2}.\text{Hz}^{-1/2}$

**Targeted Quantum projection noise floor*

from Lévêque et al. 2022

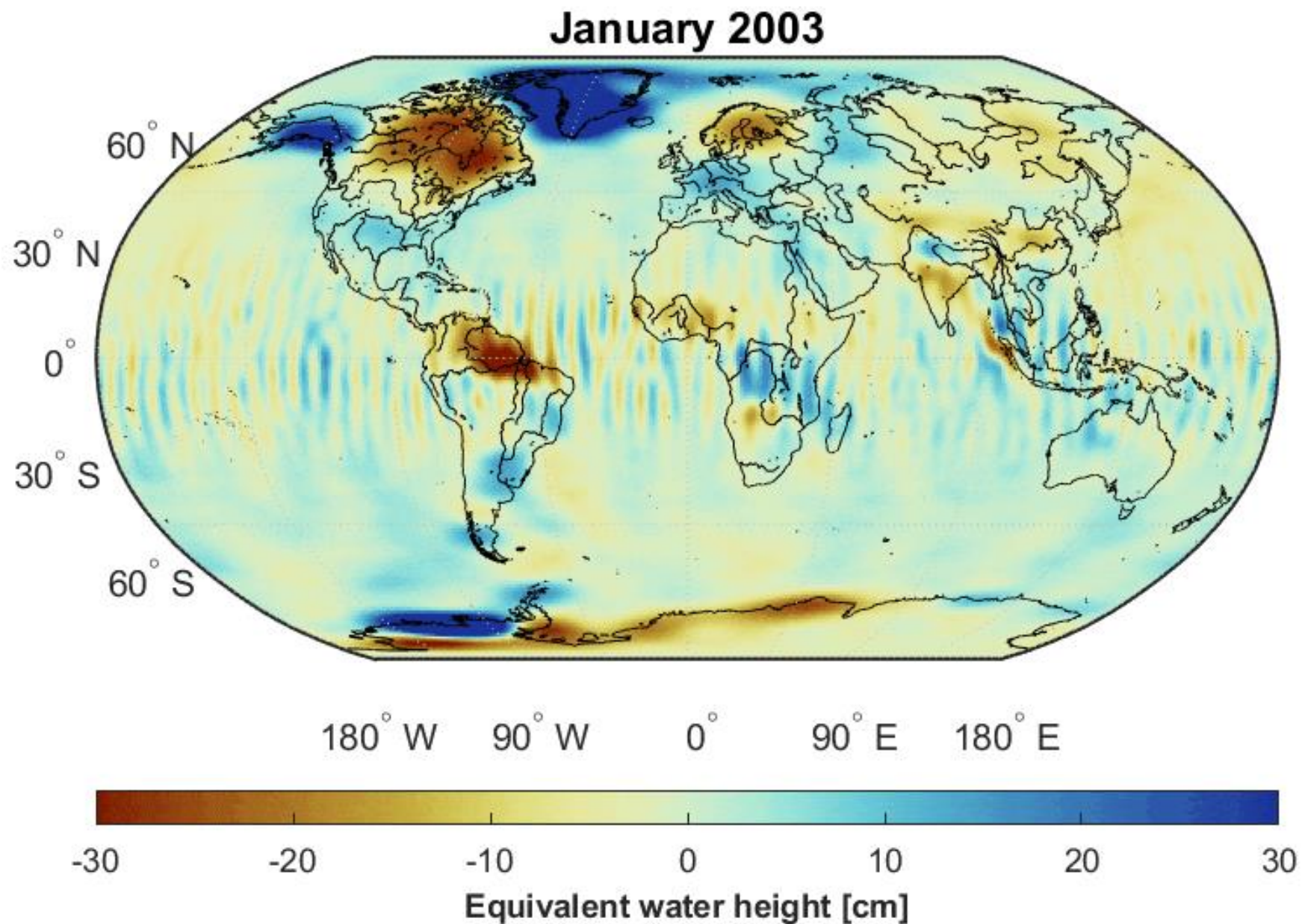
LASER INTERFEROMETRY IN SPACE

The standard approach



$$\dot{\rho} = \dot{\vec{X}}_{AB} \cdot \vec{e}_a$$

Sensing the pulse, but ...



Challenges of existing systems



SPATIAL AND TEMPORAL RESOLUTION

Limited to approx.
300km - 400km
and 1 month



SENSOR NOISE

Preventing the
full exploitation
of the ranging
systems



BACKGROUND MODELS

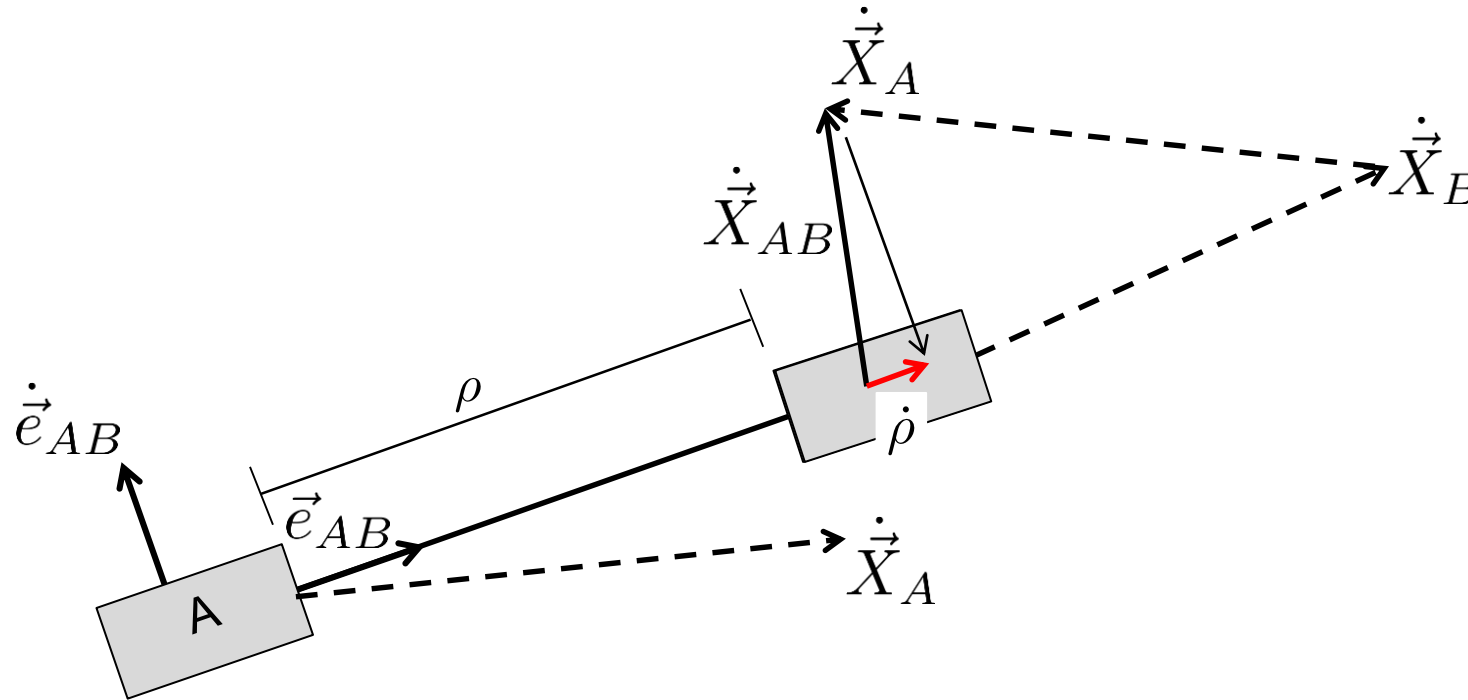
Aliasing of
unwanted geo-
physical signal



DATA AVAILABILITY

Any?
Latency?
Spatial coverage?
Sensitivity?

The second dimension ...



Rummel et al. 1978

$$\begin{aligned}\dot{\vec{X}}_{AB} \cdot \vec{e}_a &= \dot{\rho} \\ \dot{\vec{X}}_{AB} \cdot \vec{e}_r &= \rho |\dot{\vec{e}}_a|\end{aligned}$$



- Standard Approach

$$\dot{\rho} = \dot{\vec{X}}_{AB} \cdot \vec{e}_a$$

- Two-dimensional approach

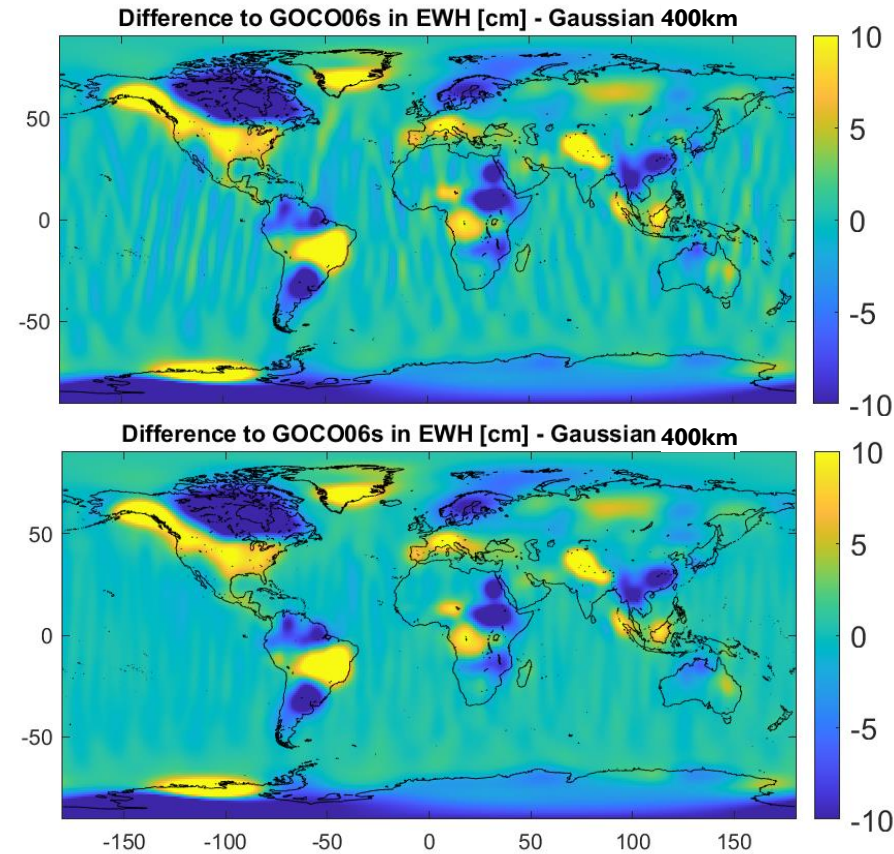
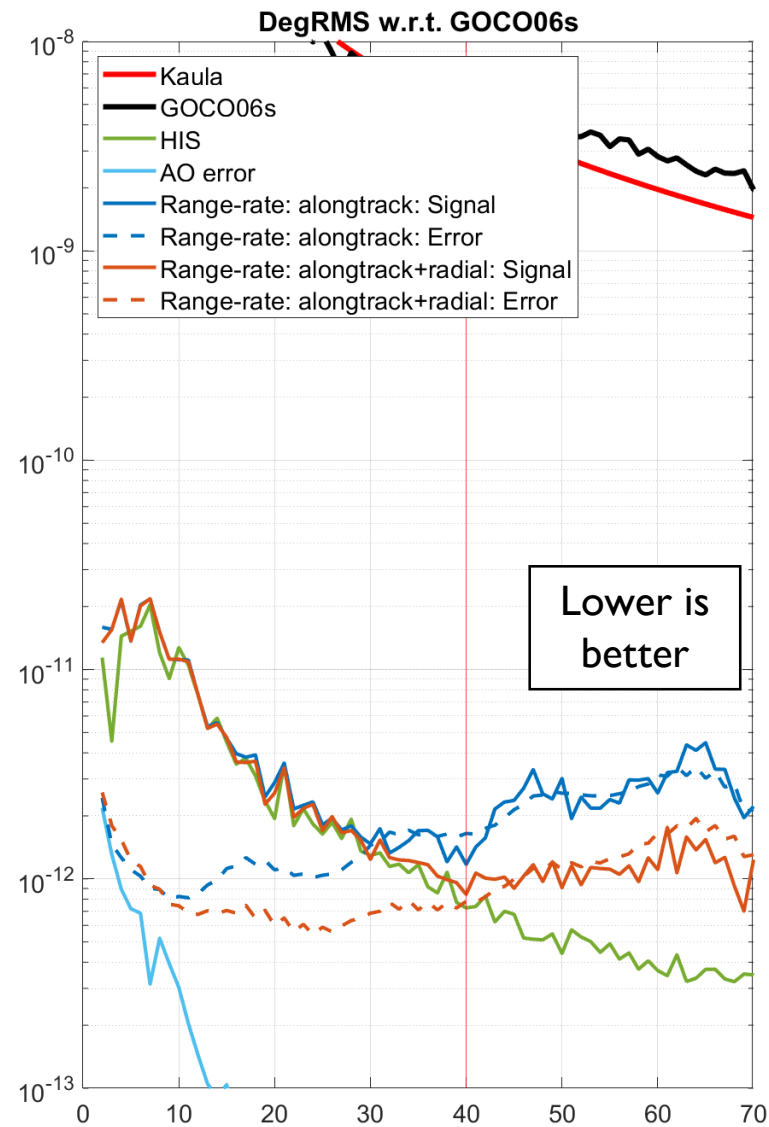
$$\dot{\vec{X}}_{AB} = \dot{\rho} \vec{e}_a + \rho \dot{\vec{e}}_a$$

introducing:

$$\vec{e}_r = \frac{\dot{\vec{e}}_a}{|\dot{\vec{e}}_a|} \rightarrow \dot{\vec{e}}_a = |\dot{\vec{e}}_a| \vec{e}_r$$

$$\dot{\vec{X}}_{AB} = \dot{\rho} \vec{e}_a + \rho |\dot{\vec{e}}_a| \vec{e}_r$$

Simulation results: 2D



Range-rate
alongtrack only

Smoother is
better

Range-rate
alongtrack+radial

The third dimension



Range observables:

$$\vec{X}_{AB} = \rho \cdot \vec{e}_{AB}^a$$

$$\dot{\vec{X}}_{AB} = \dot{\rho} \cdot \vec{e}_{AB}^a + \rho \cdot \dot{\vec{e}}_{AB}^a$$

$$\nabla V_{AB} = \ddot{\vec{X}}_{AB} = \ddot{\rho} \cdot \vec{e}_{AB}^a + 2 \cdot \dot{\rho} \cdot \dot{\vec{e}}_{AB}^a + \rho \cdot \ddot{\vec{e}}_{AB}^a$$

3D Approach:

$$\nabla V_{AB} \cdot \vec{e}_{AB}^a = \ddot{\rho} + 0 - \rho (\omega_c)^2$$

$$\nabla V_{AB} \cdot \vec{e}_{AB}^r = 0 - 2 \dot{\rho} \omega_c - \rho \dot{\omega}_c$$

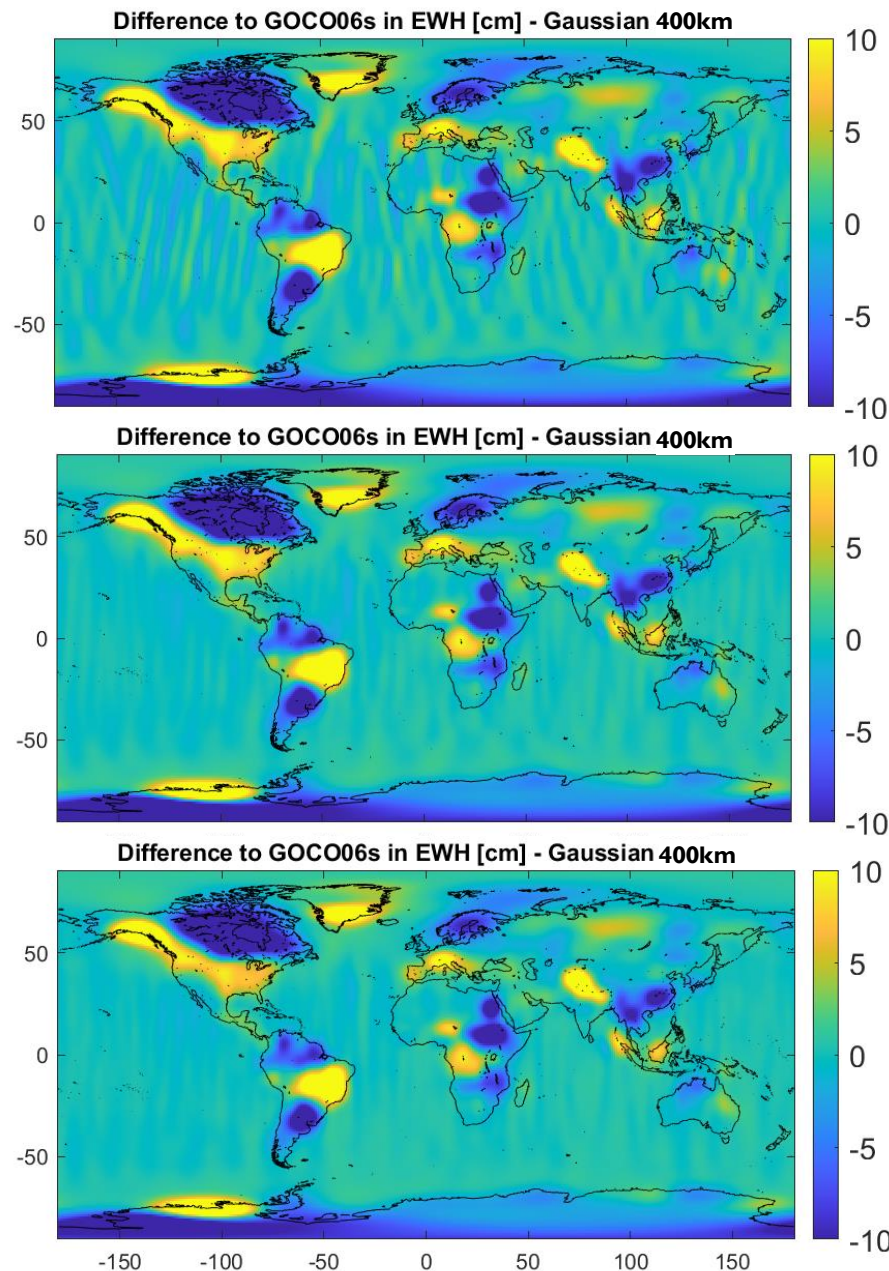
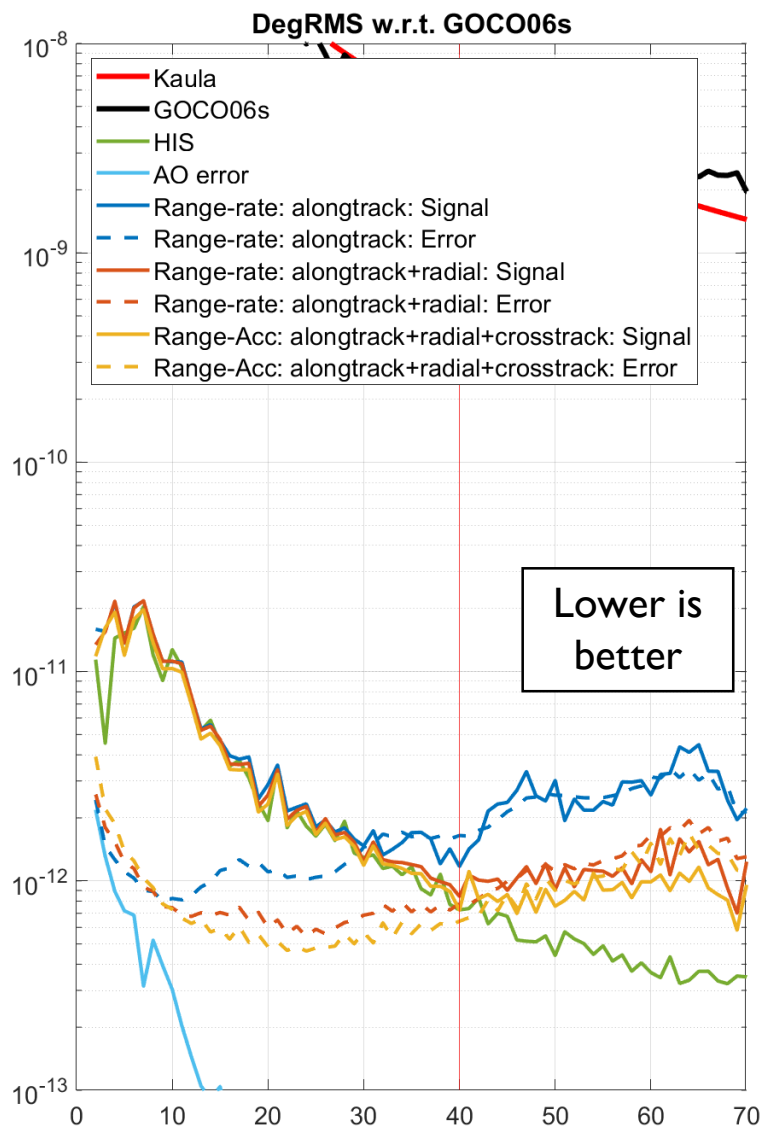
$$\nabla V_{AB} \cdot \vec{e}_{AB}^c = 0 + 0 - \rho \omega_a \omega_c$$

Weigelt 2017
*The Acceleration
Approach*

The diagram shows two spacecraft, SC1 and SC2, each with a Local Reference Interferometer (LRI) and a DWS (Dual-Wavelength System). The LOS (Line-Of-Sight) is defined by the red line connecting the LRIs of SC1 and SC2. The IF (Interferometer Frame) is defined by the blue boxes. The diagram illustrates the relationship between the LOS and the IF for both spacecraft. The LOS is shown as a red line connecting the LRIs of SC1 and SC2. The IF is shown as a blue box. The diagram illustrates the relationship between the LOS and the IF for both spacecraft. The LOS is shown as a red line connecting the LRIs of SC1 and SC2. The IF is shown as a blue box. The diagram illustrates the relationship between the LOS and the IF for both spacecraft.

A 3D rendering of the BepiColombo spacecraft in space. The spacecraft features a large, blue, hexagonal solar panel array on top, which is composed of many smaller blue panels. The main body of the spacecraft is gold-colored and has a complex, multi-faceted shape. Several silver-colored cylindrical components, likely antennas or sensors, are visible on the sides. The spacecraft is positioned diagonally against a black background filled with stars. A bright, glowing sun is visible in the upper left corner, creating a lens flare effect. In the bottom right corner, the text "©ESA/CNES" is displayed.

Simulation results: 3D



Range-rate
alongtrack only

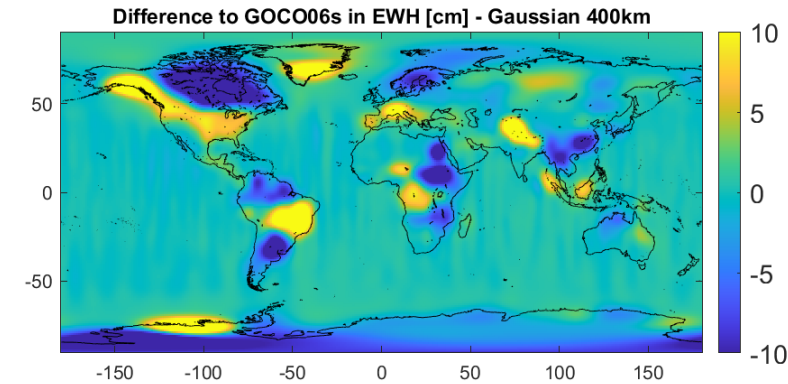
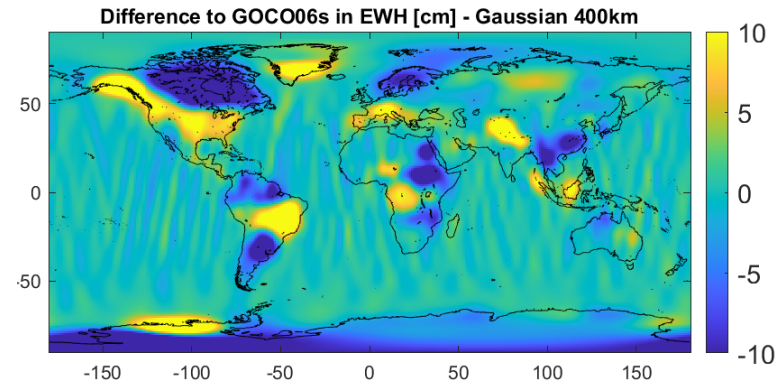
Smother is
better

Range-rate
alongtrack+radial

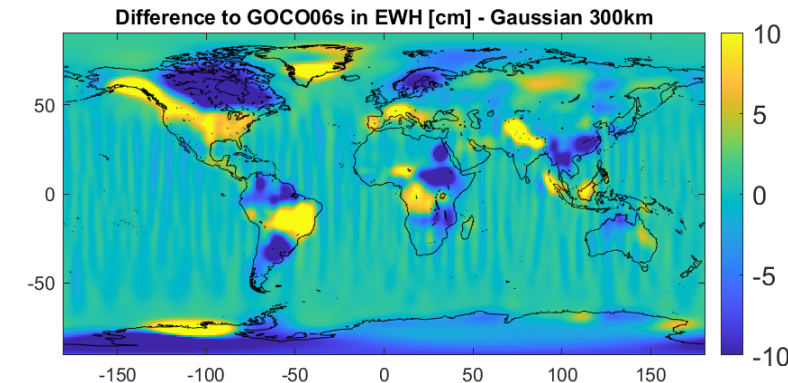
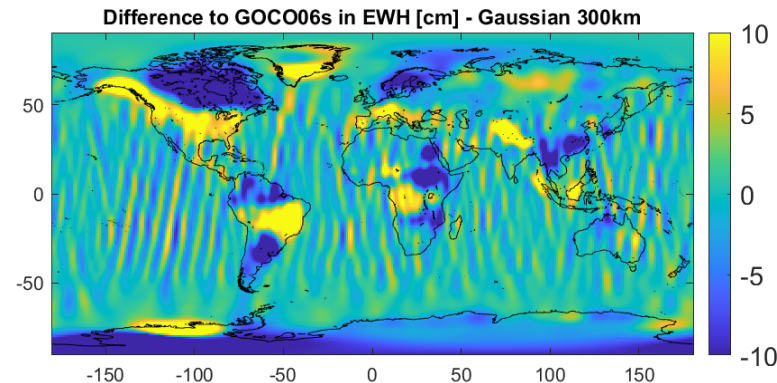
Range-acceleration
alongtrack+radial+crosstrack

Improving the spatial resolution by factor 1.5 – 2

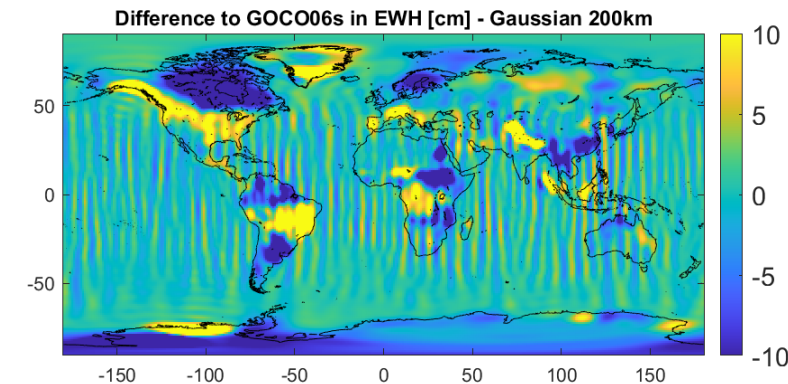
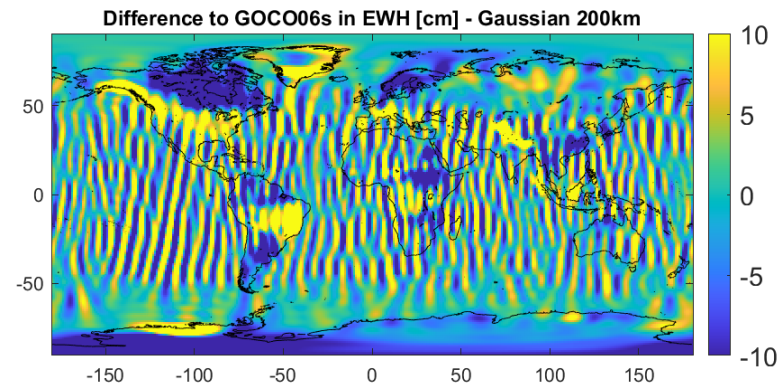
Gaussian Filtering with 400km radius



Gaussian Filtering with 300km radius



Gaussian Filtering with 200km radius



EXPLOITING LASER INTERFEROMETRY TODAY

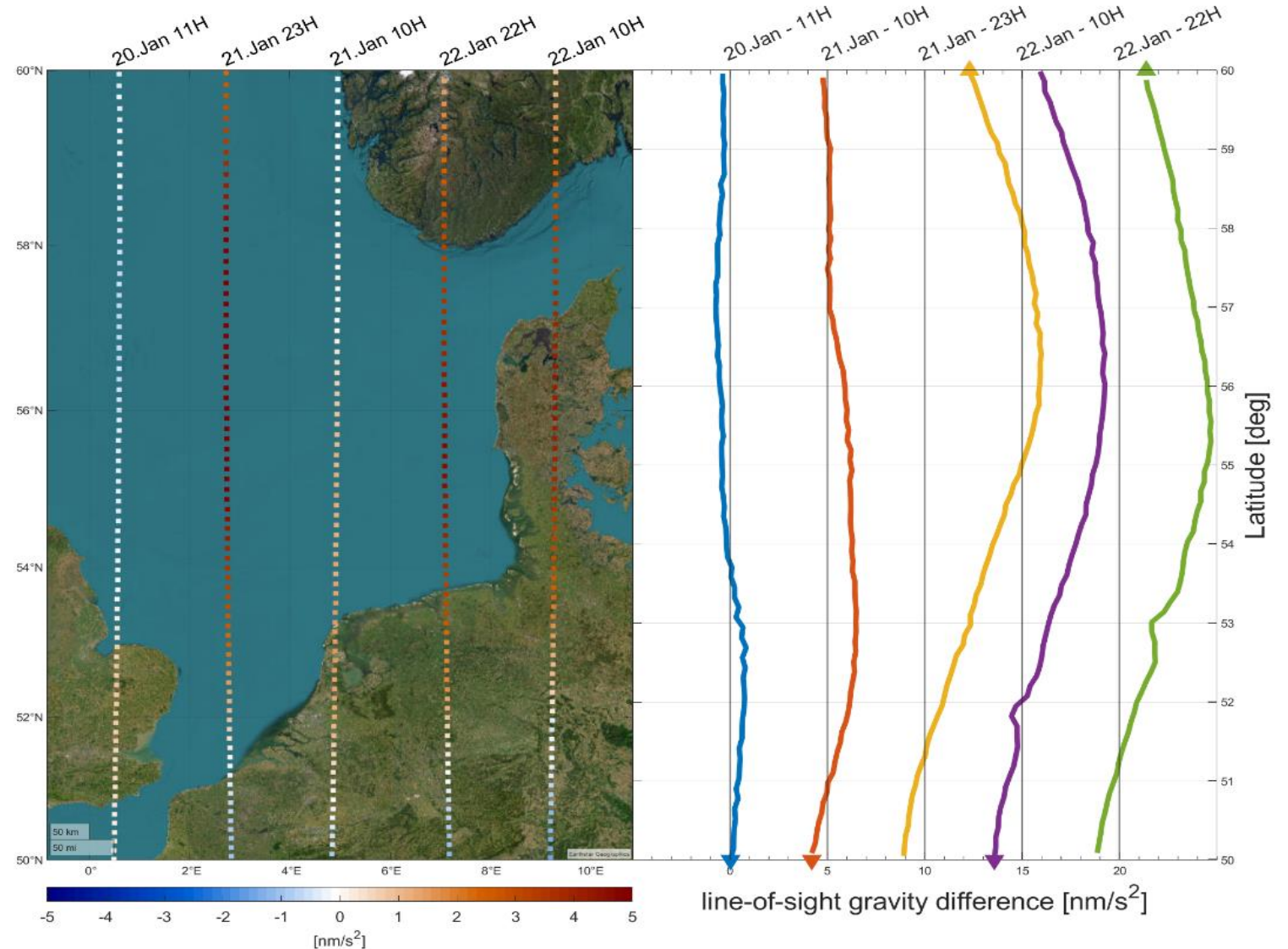


- $$\ddot{\vec{X}}_{AB} \cdot \vec{e}_{AB}^a = \Phi(\ddot{\rho})$$

Single-track analysis capabilities

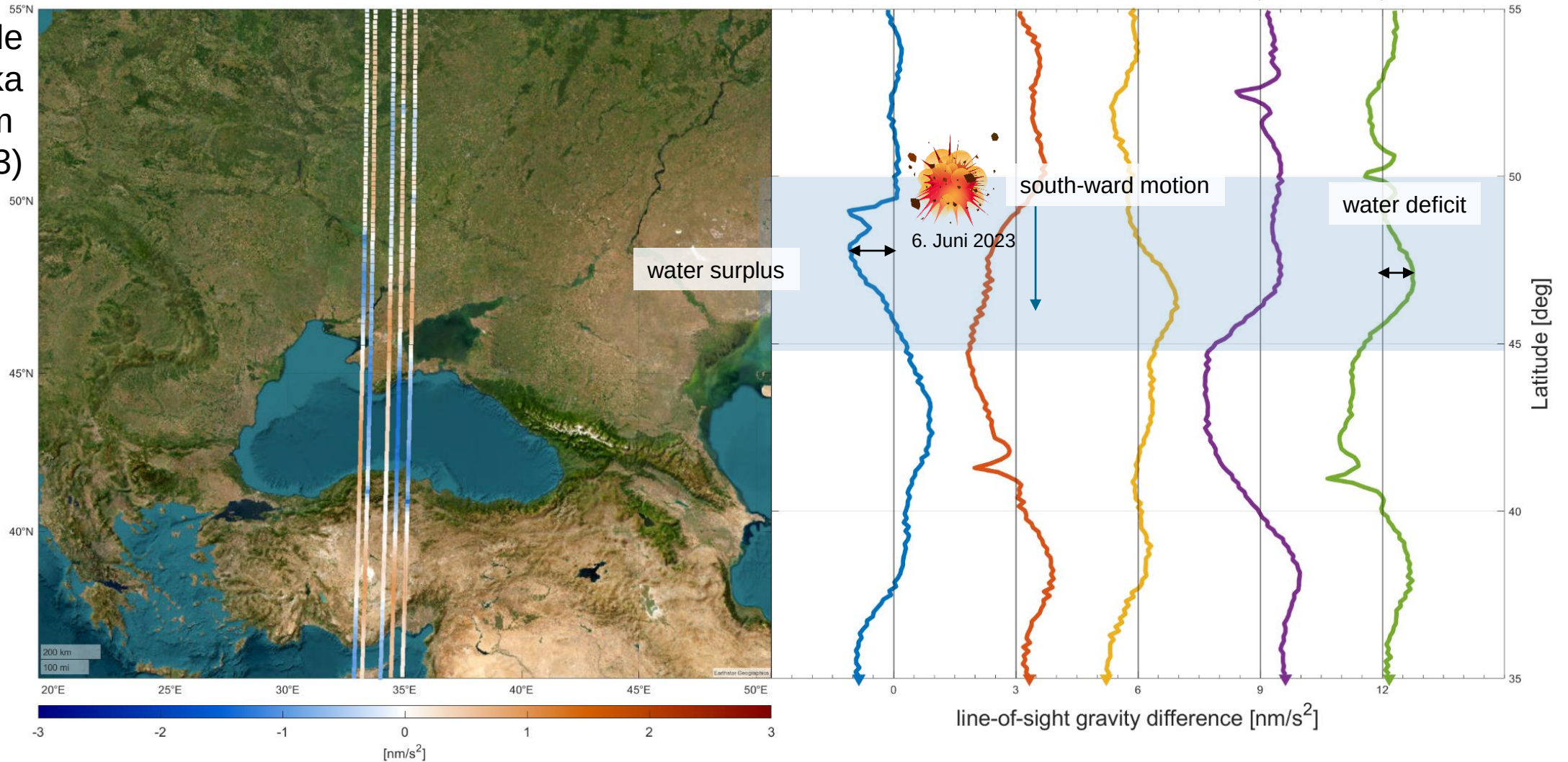
Line-of-sight
gravity differences
allows observation of
sub-monthly mass
variations

Example:
Goran
(2021)



Single-track analysis capabilities

Example
Kachowka
dam
(2023)

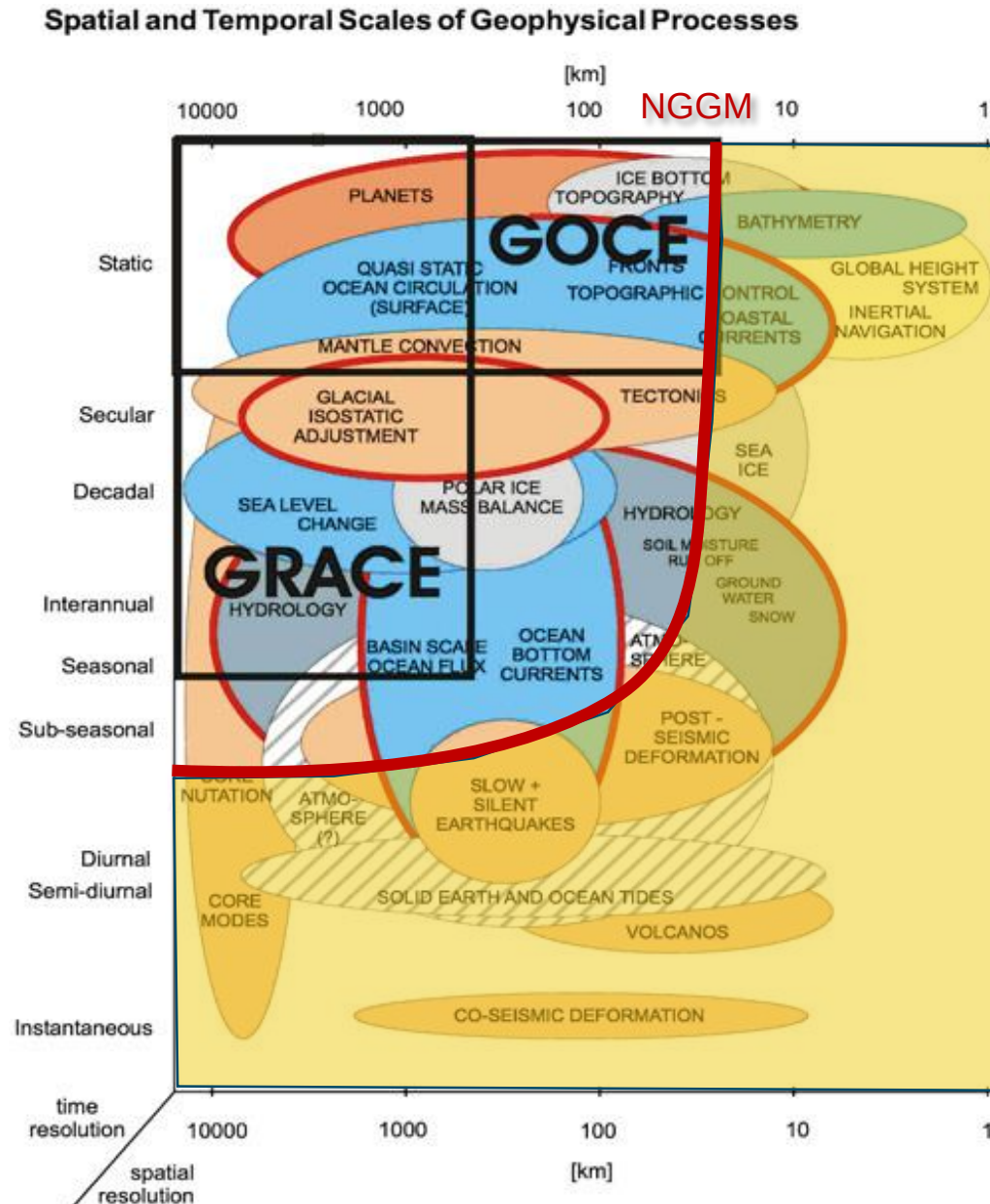




Increasing the spatial and temporal resolution

Beyond satellites

Towards sensor combination



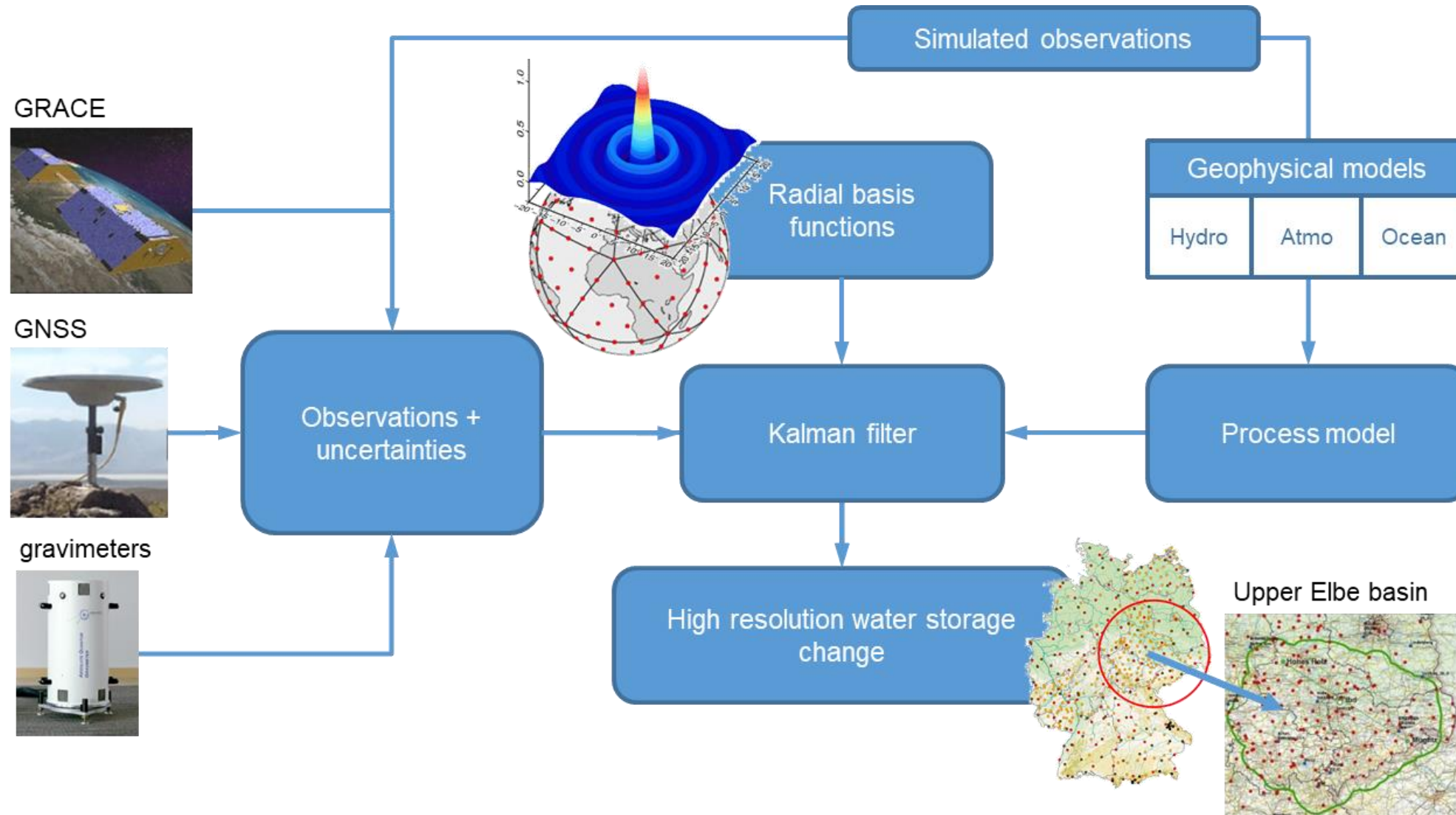
Augmentation with additional observations systems is needed.

Observation techniques:

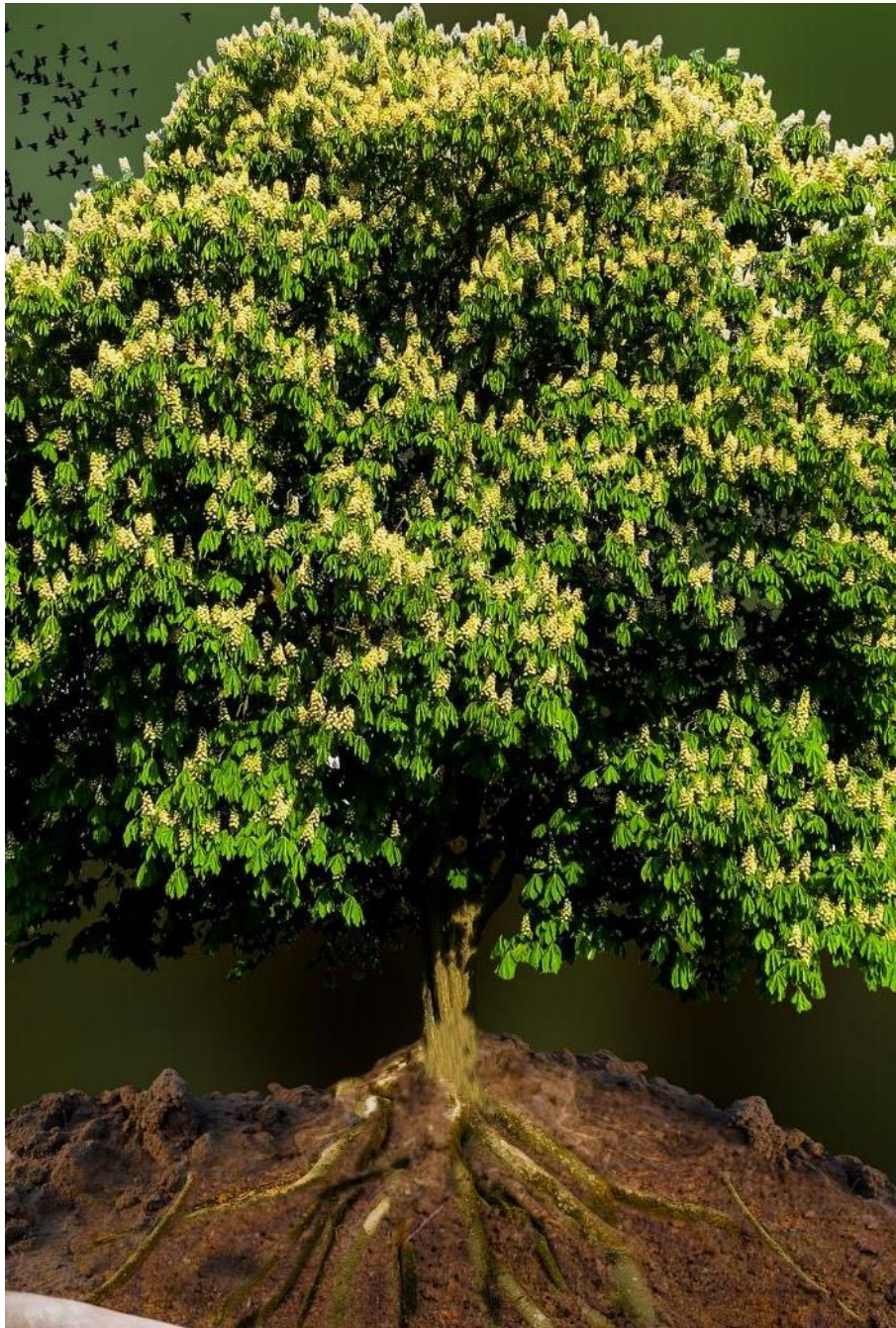
- GNSS
- InSAR
- Clocks
- Terrestrial Gravimetry

Sensor combination

Combination requires a reliable, flexible and robust estimation technique.



**Modelling of
Mass
Variations
Down to
Small Scales
by Quantum
Sensor Fusion**



Possible applications



Resource
management



Natural hazard
monitoring and
prediction



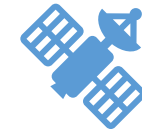
Sea-level rise
monitoring and
coastal protection



Clock-based
height reference
systems



Observation of
short-term
geophysical signals



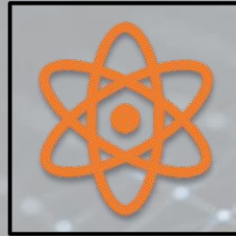
Future
satellite concepts

Fundamental objectives



Exploit

Exploit quantum technology and relativistic modelling for geodetic applications



Integrate

Integrate space and terrestrial sensors



Establish

Establish gravity field observations on all temporal and spatial scales

THANK YOU

Matthias Weigelt (matthias.weigelt@dlr.de)



Quantum sensors for interplanetary missions

- Objective: Evaluating the potential of using quantum sensors for interplanetary missions
- Adapting VENQS for Mars simulations
- Planned extensions:
 - Precise gravity field model
 - Tides and 3rd body forces
 - Atmospheric drag
 - Solar radiation pressure + Albedo

